

# Discussion of Open Issues on Cracks Versus Notches

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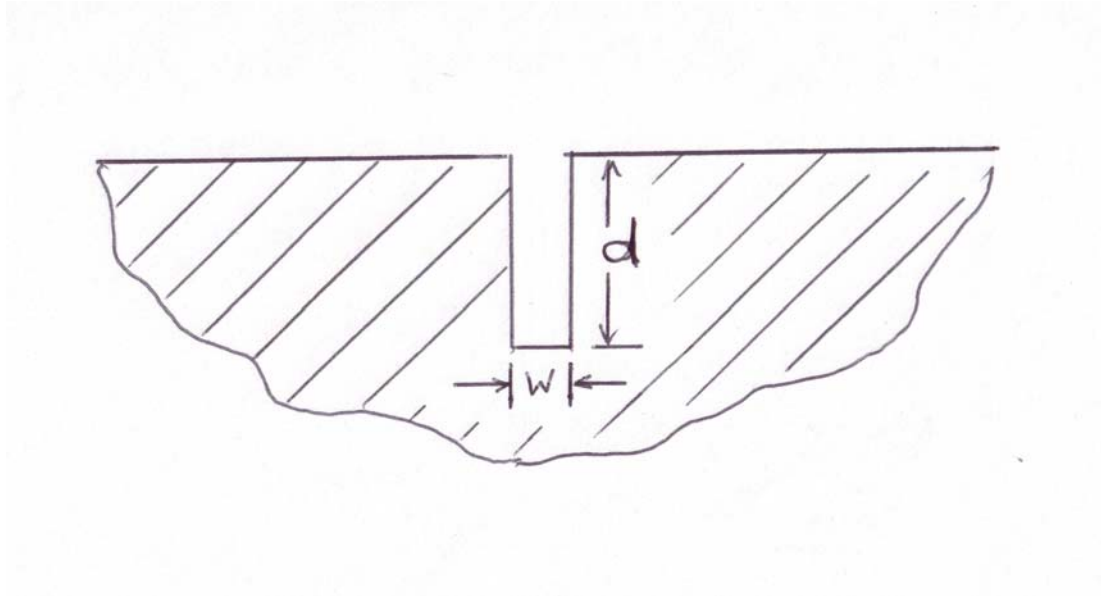
# Outline

- Slides from February, 2005 MAPOD Meeting
- Controlling Factors
- Past Eddy Current Data
- Past Ultrasonic Data

# Outline

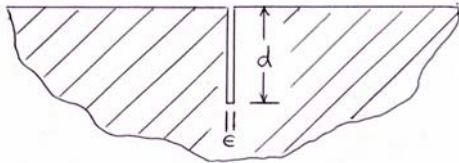
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# Notch

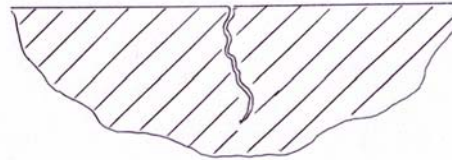


# Cracks

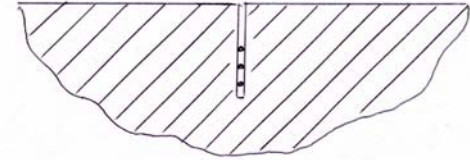
Ideal Mathematical Crack



Morphology Effects



Electrical/Mechanical Contact Effects



Material Mechanisms

- Growth along grain boundaries
- Non-uniform residual stresses

- Oxides and other debris
- Contacting asperities
- Sheared faces

# Ultrasonics

Response as Compared to Notch Response

<b>Measurement</b>	<b>Ideal Crack</b>	<b>Morphology Effects</b>	<b>Mechanical Contact Effects</b>
<b>Specular Reflection</b>	Equivalent	Reduced Due to Interference	Reduced Due to Transmission
<b>Tip Diffraction</b>	Different; Often Less	Different; Often Less	Different; Often less
<b>Through Transmission</b>	Equivalent	Equivalent	Increased

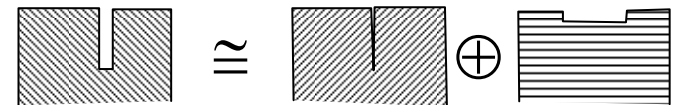
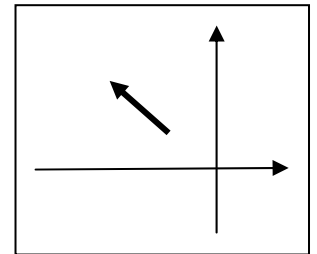
# Eddy Currents

## General Comments

- Electrical contacts (bridging) will always have an effect if currents, following along crack faces, are “short circuited”
- Morphology effects are less significant than for UT
- Open cracks have greater “inductance” than ideal mathematical crack because of stored energy in magnetic fields
- The difference increases with frequency
- In the impedance plane, this is similar to, and hard to differentiate from, lift-off effect

# Notch vs. Crack: EC Model

- Notch-Crack difference appears
  - Strongly in impedance amplitude
  - Weakly in vertical components (when lift-off is horizontal)
- Reason
  - The volume effects behave similarly to the lift-off effect
    - More volume energy = higher reactance
    - Less material = lower resistance





# Example Calculation

## Model Parameters

### ■ Notch *length* × *depth* × *width*

- $l=1\text{mm}$ ,  $d=0.5\text{mm}$
- $w=0.0, 0.05, 0.1\text{mm}$

### ■ Solenoid coil

- $ID=1.07\text{mm}$ ,  $OD=2.62\text{mm}$
- $L=2.79\text{mm}$
- Lift off= $0.73\text{mm}$
- $F=100\text{kHz}$

### ■ Part = a plate

- Inconel 600 ( $1.02 \times 10^6 \text{ S/m}$ )
- 1.27mm thick

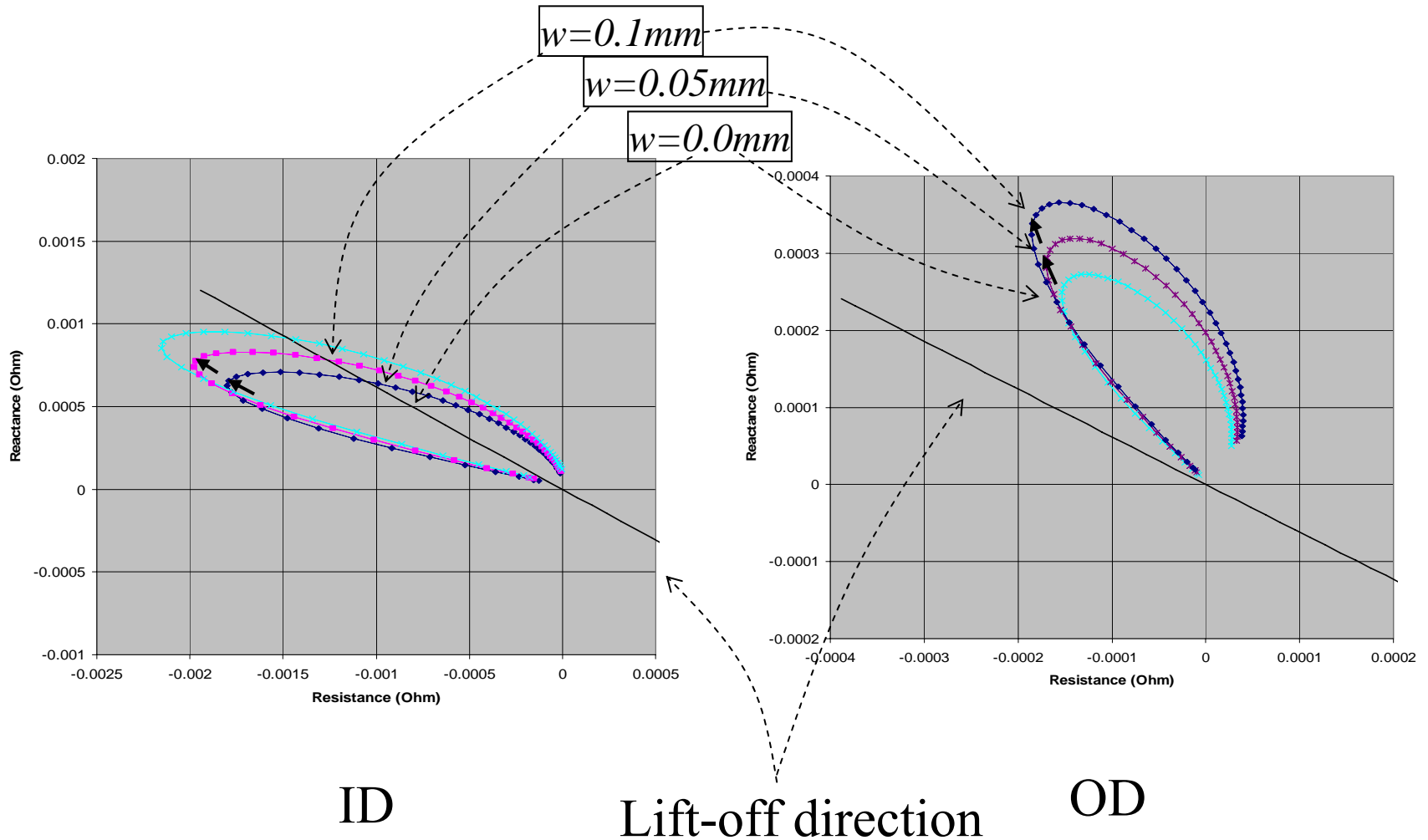
### ■ In two configurations

- “ID” (same side)
- “OD” (opposite side)

### ■ Results

- ~20% increase in amplitude with 10% opening (i.e.  $w/l=0.1$ )
- Increase in the lift-off direction
- Vertical components are insensitive to notch openings.

# Calculated Opening Effects



# Eddy Currents

<b>Measurement</b>	<b>Ideal crack response as compared to notch response</b>
<b>Absolute coil</b> <ul style="list-style-type: none"><li>■ Lift-off rotated to horizontal</li><li>■ “Response” taken as vertical response</li></ul>	<b>Difference often small</b> <ul style="list-style-type: none"><li>■ Ideal crack can have greater or less response</li></ul>
<b>Differential coil</b> <ul style="list-style-type: none"><li>■ “Response” taken as magnitude of impedance change</li></ul>	<b>Significant Difference</b> $ \Delta Z_{NOTCH}  >  \Delta Z_{CRACK} $

# Suggested Strategy

- Use physics-based models to correct notch data for difference between ideal cracks and notches
- Create database of deviations of responses real cracks from expectations for ideal cracks
  - Include salient materials variables specifying growth factors controlling morphology
    - HCF vs LCF
    - Closure
    - Etc.
- Long term goal
  - Develop “knock down factors” that can be confidentially used in new studies

# Outline

- Slide from February, 2005 MAPOD Meeting
- **Controlling Factors**
- Past Eddy Current Data
- Past Ultrasonic Data

# Controlling Factors for Signal – EC (to be checked against INCITE list)

- Correlation between signal and noise sources
- Notches
  - Wire
  - HAZ
  - Tool condition
  - Processes to generate (feeds, speeds)
  - Tool materials and dimensions
  - Tool shape
  - Notch shape

# Controlling Factors for Signal – EC (to be checked against INCITE list)

## ■ Cracks

- Opening
- Contacting asperities
- Roughness
- Surface condition (shot peening, smeared metal)
- Presence of fretting
- Morphology (shape, orientation, depth, length)
- Multiple cracks vs. single crack

# Controlling Factors for Signal – EC (to be checked against INCITE list)

## ■ Root causes

### □ Growth conditions

- $\sigma/\sigma_y$  time history
  - Constant amplitude
  - Overloads
  - High or low cycle
  - Mode I, II, III or ?

### □ Initiation conditions

- Stress corrosion vs. fatigue vs. corrosion fatigue, etc.
- Intergranular vs. transgranular
- Scratches, dings
- Fretting
- Surface vs. subsurface initiation



# Controlling Factors for Signal – EC (to be checked against INCITE list)

## □ Material issues

- Toughness
- Grain size
- Grain boundary condition
- Mechanism of contact (sliding, oxide debris, plastic deformation)
- Contacts conducting
- Roughness

# Controlling Factors for Noise - EC

- Scratches, dents, dings
- Roughness
- Surface geometry features (edges, corners, etc.)
- Out of roundness
- Corrosion, pitting
- Dirt
- Liftoff variations
- Microstructure
- Thermal drift
- Fastener/part interface and material

# System/Operator Issues

Consider relevance to model inputs some of which will be modeled and some of which will be characterized empirically

- Probes, instrumentation, cables, etc.
- Scan plan (speed)
- Quality control of inspection system
- Human factors

# Challenge: Variable for Fastener Sites (Aldrin, et. al)

- NDE technique (measurement system):
  - NDE method
  - Transducer/probe design
  - Contact condition with part (direct, immersion)
  - Scan plan (directions, resolution, orientation)
- Part Geometry, material and condition:
  - Layer material, number, and thickness (shims)
  - Outer layer surface condition (paint, corrosion)
  - Fastener material/type/head condition

# Challenge: Variable for Fastener Sites (Aldrin, et. al)

- Hole geometry (oblong, off-angled, surface conditions, scratches, chatter, tool marks)
- Fastener hole fit (asymmetric fit, irregular contact conditions/loading, sealant)
- Gaps/sealant between layers (aging)
- Presence of metal shavings
- Bushings, residual stress around holes
- Proximity of adjacent fasteners and edges
- Presence and condition of repairs

# Challenge: Variable for Fastener Sites (Aldrin, et. al)

## ■ Flaw characteristics:

- Flaw number (number of cracks per site)
- Flaw type (cracks, EDM notch)
- Flaw location (layer, location in layer: surface, mid-bore, eye-brow cracks)
- Flaw orientation (around fastener site, skew angle from normal)
- Flaw dimensions (length, aspect ratio)
- Material within flaw (sealant/paint/fluids)
- Flaw morphology (regular, irregular)
- Flaw conditions at crack faces (contact conditions, residual stress)

# Controlling Factors for Signal – UT (surface breaking cracks)

- Correlation between signal and noise sources
- Notches
  - Wire
  - HAZ
  - Tool condition
  - Processes to generate (feeds, speeds)
  - Tool materials and dimensions
  - Tool shape
  - Notch shape

# Controlling Factors for Signal – UT (surface breaking cracks)

## ■ Cracks

- Opening
- Contacting asperities
- Roughness
- Surface condition (shot peening, smeared metal)
- Presence of fretting
- Morphology (shape, orientation, depth, length, branching)
- Multiple cracks vs. single crack



# Controlling Factors for Signal – UT (surface breaking cracks)

## ■ Root causes

### □ Growth conditions

- $\sigma/\sigma_y$  time history
  - Constant amplitude
  - Overloads
  - High or low cycle
  - Mode I, II, III or ?

### □ Initiation conditions

- Stress corrosion vs. fatigue vs. corrosion fatigue, etc.
- Intergranular vs. transgranular
- Scratches, dings
- Fretting
- Surface vs. subsurface initiation
- Multiple indications in the area of interest

# Controlling Factors for Signal – UT (surface breaking cracks)

## ■ Material Issues

- Toughness
- Grain size
- Grain boundary condition
- Mechanism of contact (sliding, oxide debris, plastic deformation)
- Contacts conducting
- Roughness

# Controlling Factors for Noise - UT

- Scratches, dents, dings
- Roughness
- Surface geometry features (edges, corners, etc.)
- Out of roundness
- Corrosion, pitting
- Dirt
- Micro/macro structure (anisotropy, attenuation)
- Sealant variation, bladders, foam
- Surface protectants (paint, coatings, etc.)
- Couplant variation
- Interface contaminants
- Thermal drift
- Fastener/part interface and material

# System/Operator Issues

Consider relevance to model inputs some of which will be modeled and some of which will be characterized empirically

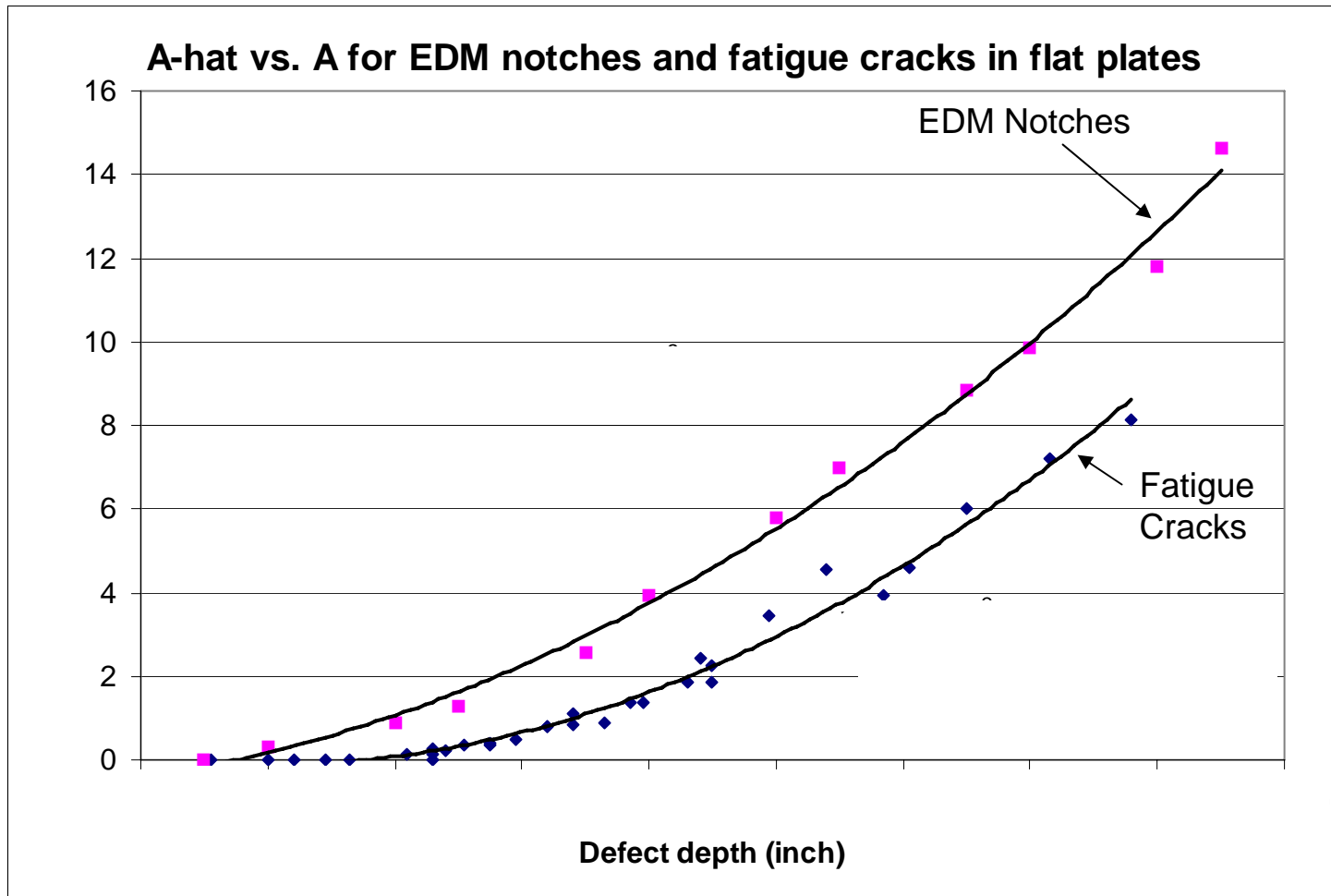
- Probes, instrumentation, cables, etc.
- Scan plan (speed)
- Quality control of inspection system
- Human factors

# Outline

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- **Past Eddy Current Data**
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# Transfer Function Example

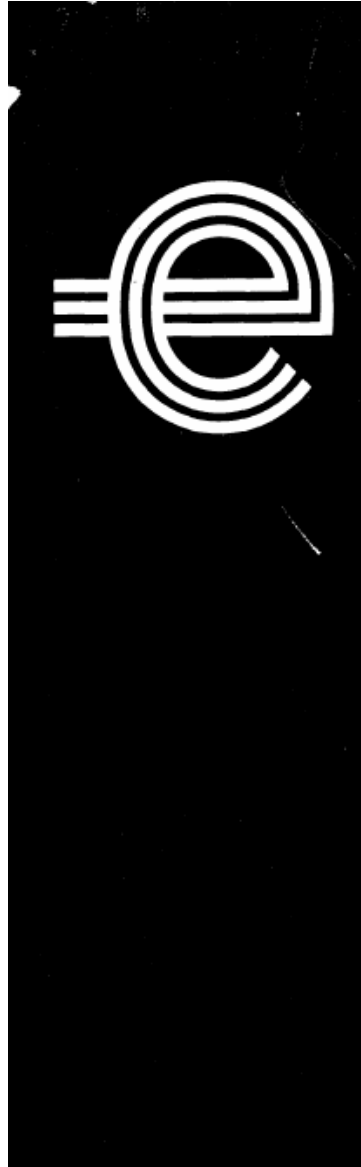
From Smith (Pratt & Whitney) – February 2005 MAPOD Meeting



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# Fatigue Crack Growth and Compressive Stress



Report No. NW/SSD/RR/45/80

Central Electricity Generating Board  
North Western Region  
Scientific Services Department

R40485

THE INFLUENCE OF CRACK GROWTH CONDITIONS AND  
COMPRESSIVE STRESS ON THE ULTRASONIC DETECTION  
AND SIZING OF FATIGUE CRACKS.

by

A.B. Wooldridge and G. Steel

Date: April 1980



# Fatigue Crack Growth and Compressive Stress

## ■ Summary

- A series of fatigue cracks in mild steel parent metal and weld metal grown under constant stress intensity conditions have been examined ultrasonically for compressive stresses up to  $150 \text{ MN m}^{-2}$ . Various angles of shear waves and the Delta technique were employed to study the corner echoes. Reductions in reflectivity at zero load and under stress have been shown to correlate with the crack growth conditions and with the roughness of the crack faces. We also measured the ultrasonic echoes from the crack tips which are small even at zero load and become undetectable for small compressive stresses. The detectability with shear waves of cracks containing liquid has been measured and compared with theoretical predictions derived from a thin parallel-sided gap model.

# Fatigue Crack Growth and Compressive Stress

## ■ Conclusions

- The growth conditions of fatigue cracks have a significant affect on their ultrasonic response, both at zero load and when under compressive stress. The cyclic change in stress intensity factor during crack growth correlates well with the roughness of the fatigue crack surfaces and this is believed to cause the changes in ultrasonic response.
- Both increasing crack roughness and increasing compressive stresses reduce the specular reflection from cracks but the roughest cracks show the least variation with stress.

# Fatigue Crack Growth and Compressive Stress

## ■ Conclusions

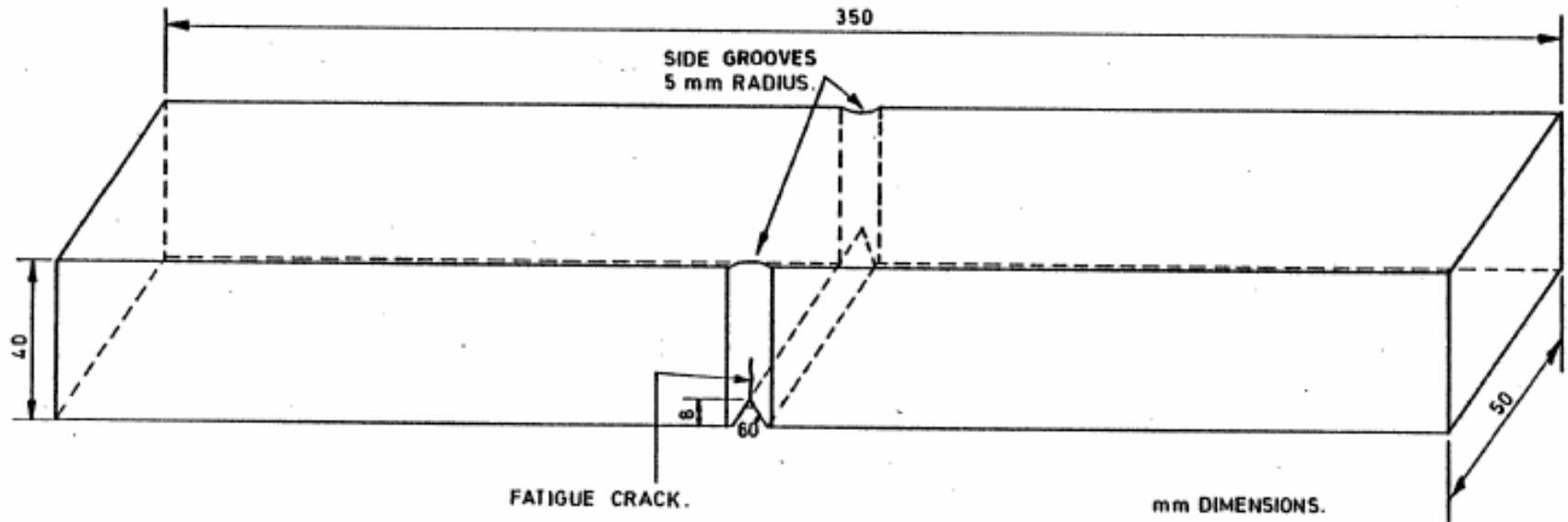
- Crack tip echoes are small; typically they are 50 dB down on a back wall echo at the same range when using a compression wave probe at grazing incidence, and they are practically impossible to identify reliably if the cracks are in compression or if the material contains other defects such as inclusions.
- The errors in sizing cracks in clean material by detecting the tip echoes are typically  $\pm 1$  mm if averaged for several probes. Individual readings, however, may be in error by several millimeters.

# Fatigue Crack Growth and Compressive Stress

## ■ Conclusions

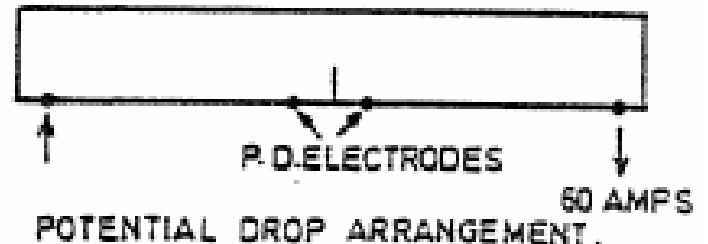
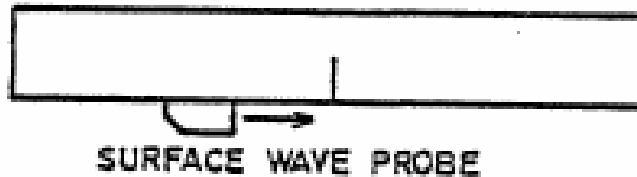
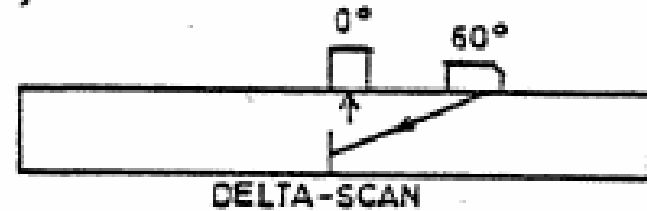
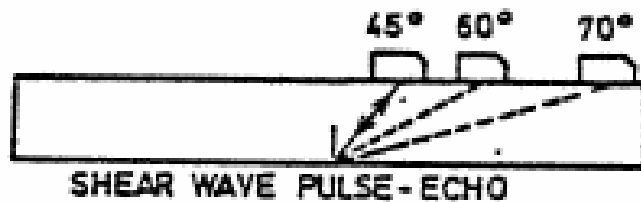
- The presence of liquid in a crack causes a marginal increase in reflection for shear wave beams incident at  $20^\circ$  to the crack normal. Modest decreases in reflection occur for beams incident at  $45^\circ$ , while considerable decreases are likely at  $30^\circ$  incidence.

# Fatigue Crack Growth and Compressive Stress

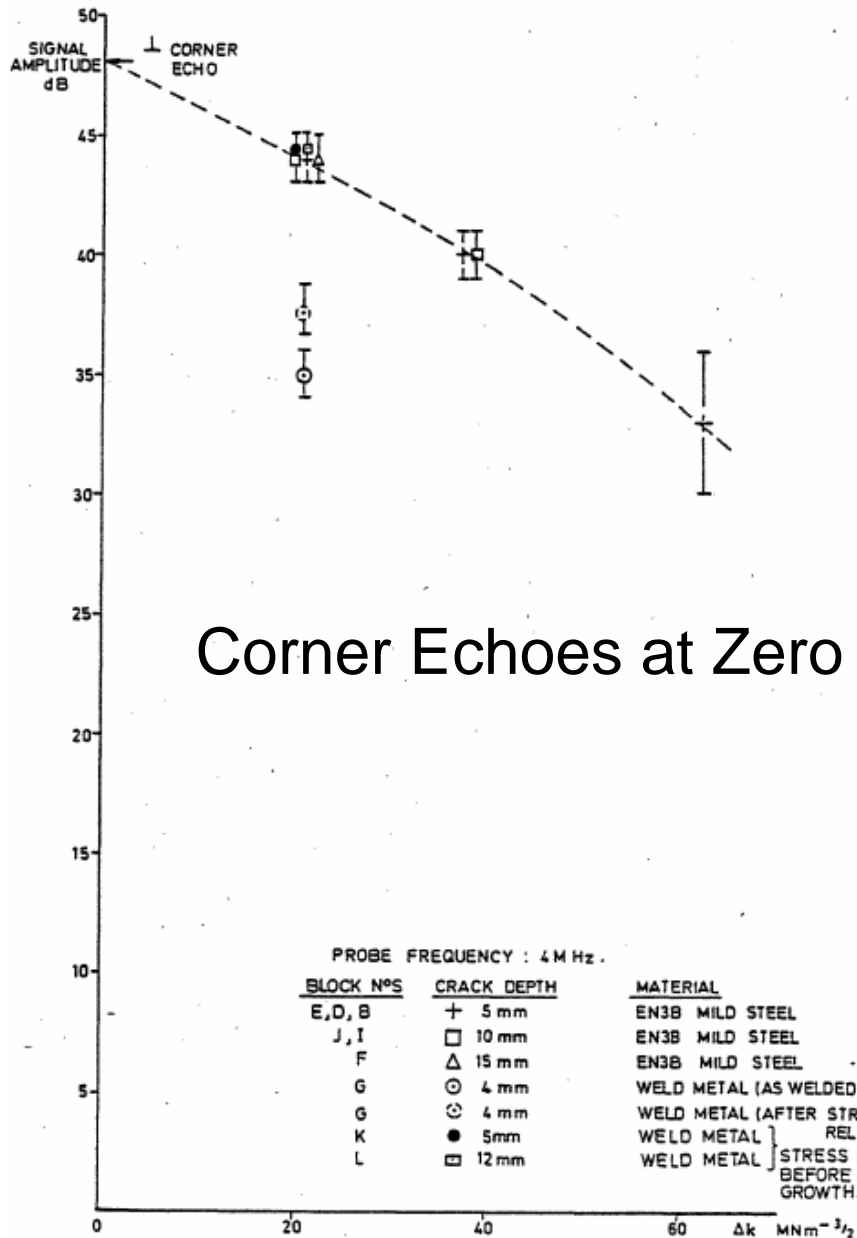


Initial Fatigue Crack Specimen

# Fatigue Crack Growth and Compressive Stress

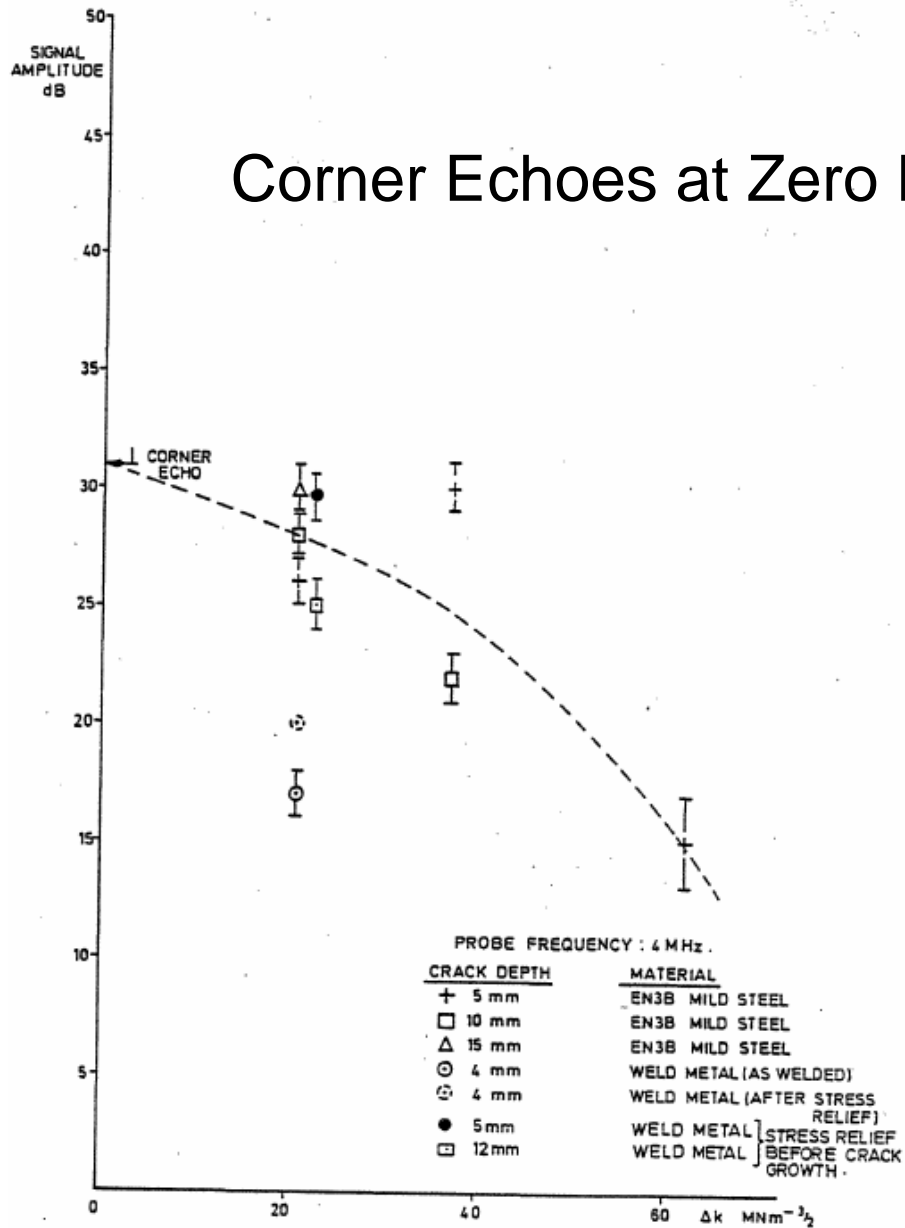


Ultrasonic Probe Arrangements

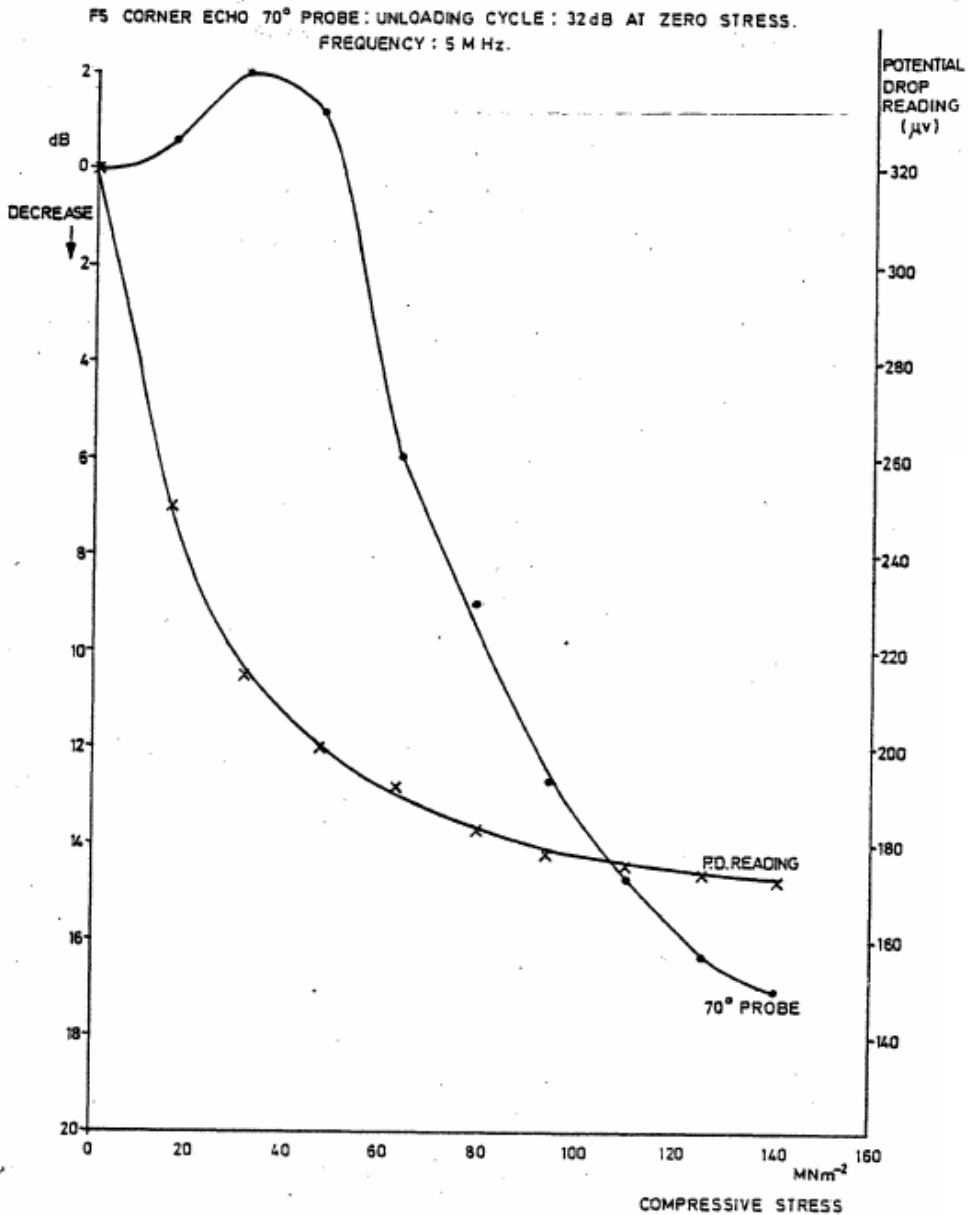


## Corner Echoes at Zero Loads Versus $\Delta k$ : 45° Probe

# Corner Echoes at Zero Loads Versus $\Delta k$ : 60° Probe

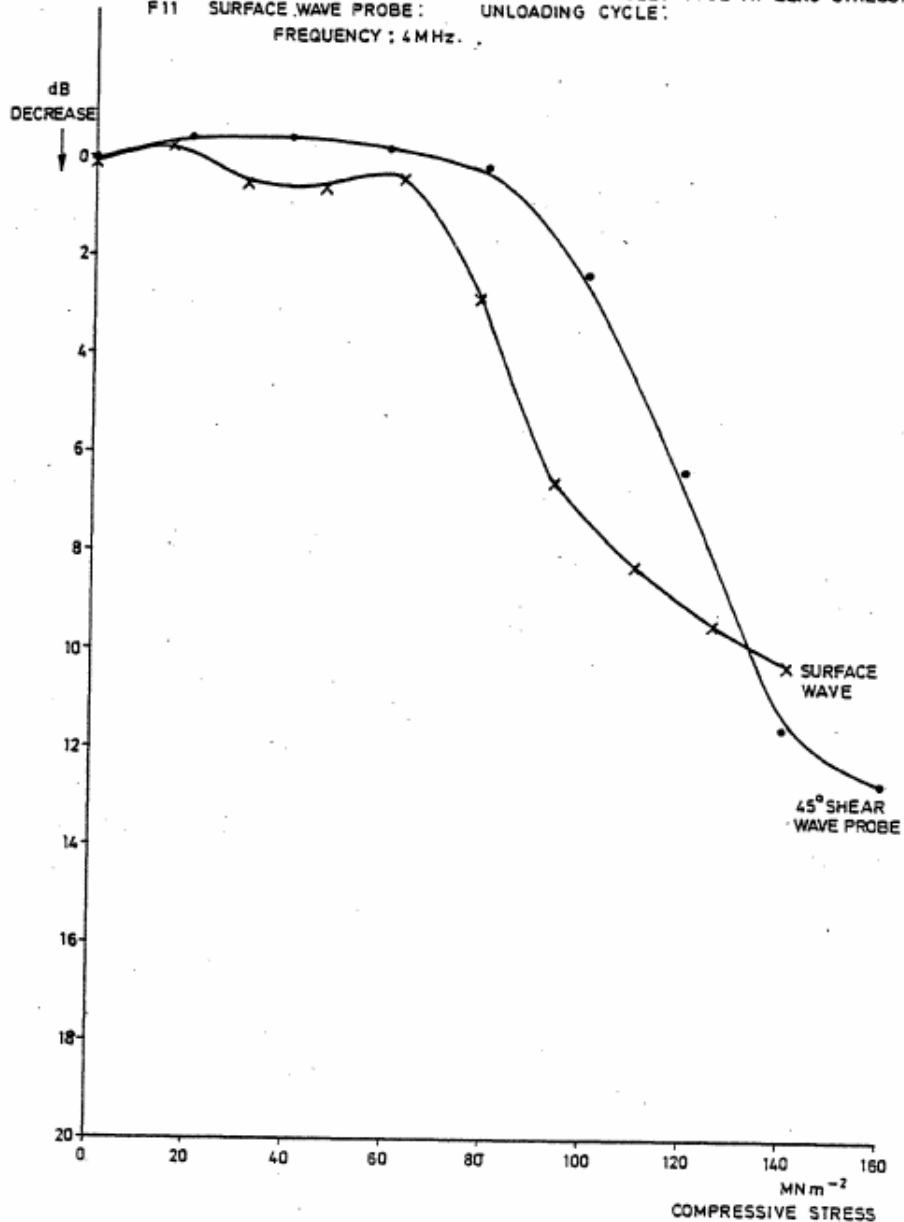




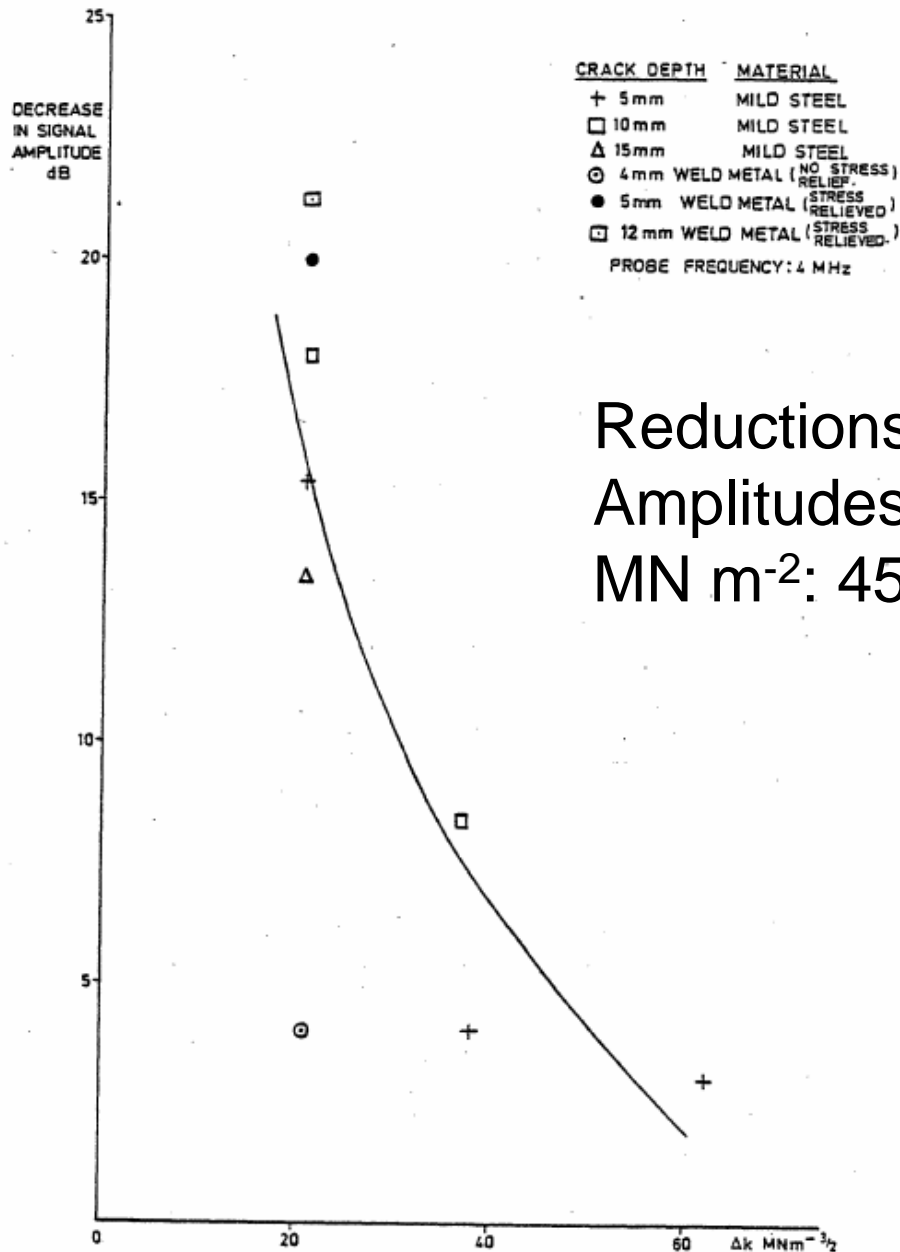


## Block F: 70° Shear Wave Probe Echo and Potential Drop Reading

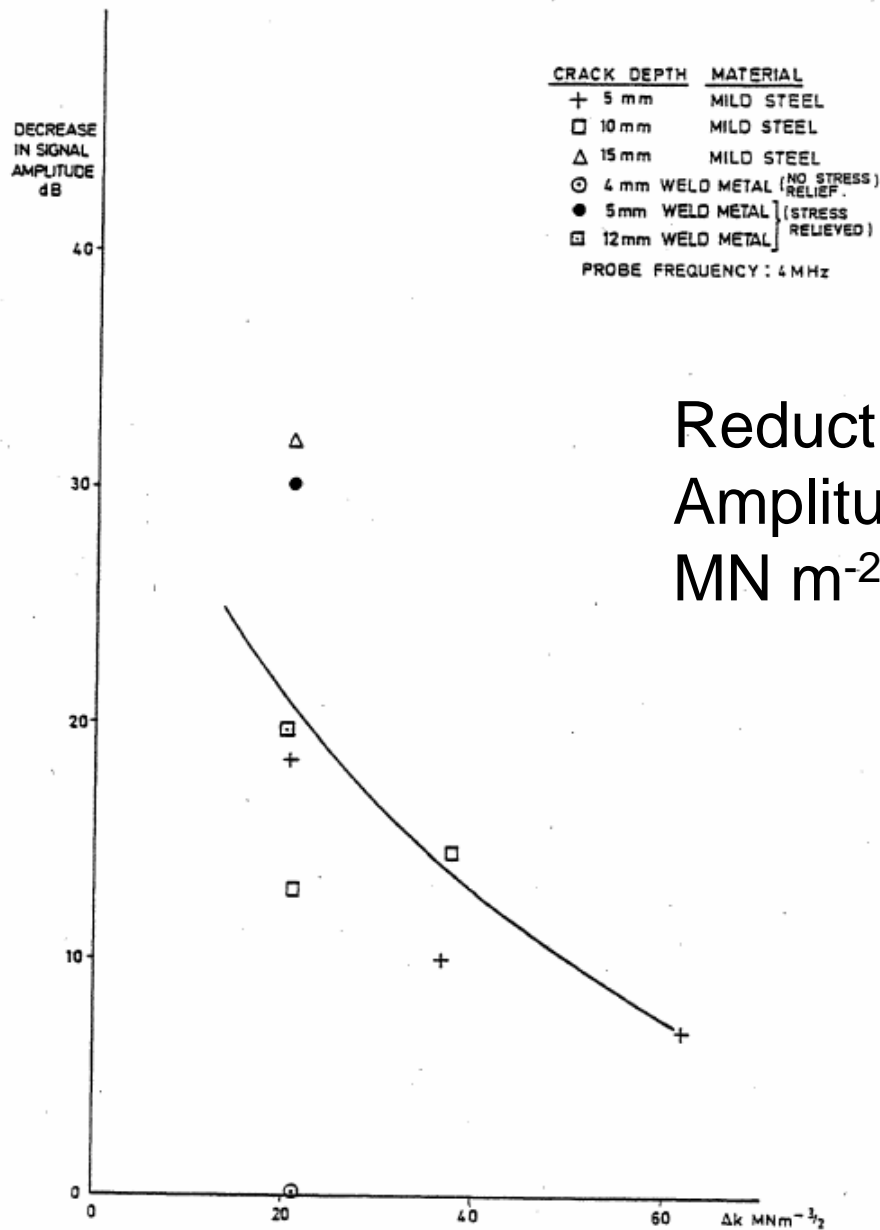
F2b CORNER ECHO 45° PROBE: UNLOADING CYCLE: 44dB AT ZERO STRESS.  
F11 SURFACE WAVE PROBE: UNLOADING CYCLE:  
FREQUENCY: 4MHz.



## Block F: 45° Shear Wave and Surface Wave Echo



Reductions in Corner Echo Amplitudes for a Stress of 160 MN m<sup>-2</sup>: 45° Probe



Reductions in Corner Echo Amplitudes for a Stress of 160 MN m<sup>-2</sup>: 60° Probe