Agency Perspective – AFRL

Vision of the Need for Model-Assisted Approaches to POD Determination



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Inputs were gathered from several people in the AF aging aircraft community:

- Joseph Gallagher
- Mike Paulk
- Bob Lewis
- Charlie Buynak
- Gary Steffes
- Matt Golis
- Tom Moran



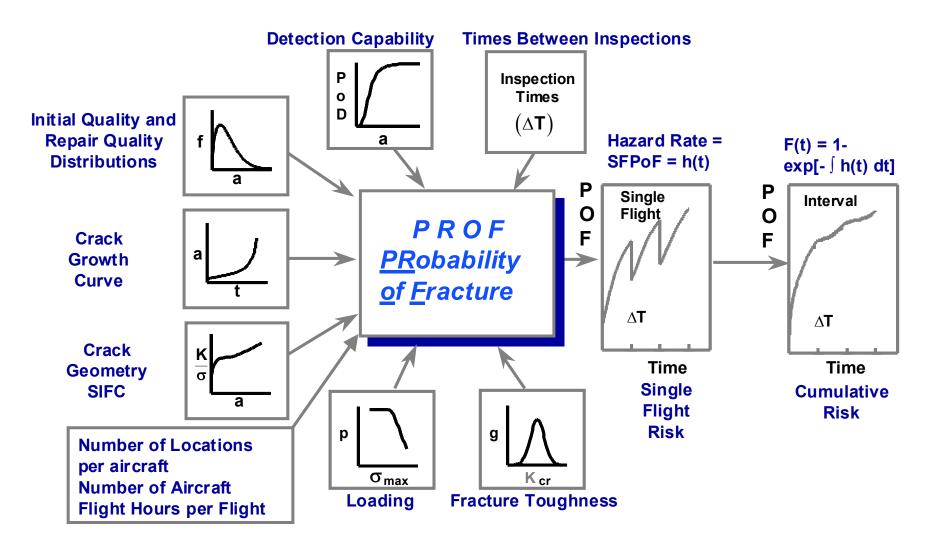
Wave of Requirements and Technologies



- Man-hours for NDT scheduled to increase dramatically!
- Need to insert new technologies into the field, faster and cheaper!
- Implementation of inspections without POD undermines NDE!
- Damage tolerant risk analysis techniques demand Quantitative NDE! (Gallagher, Babish, and Malas, 2005)

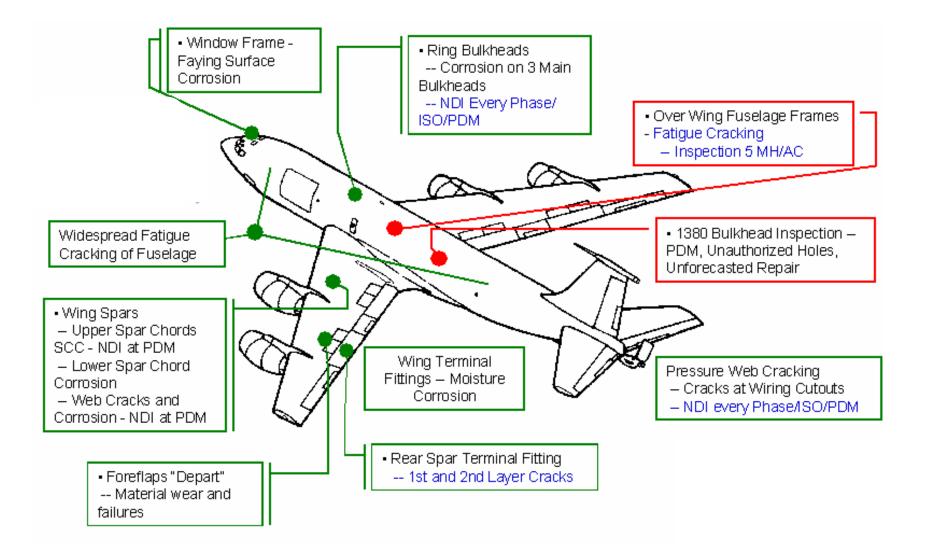


Risk Analysis Input parameters



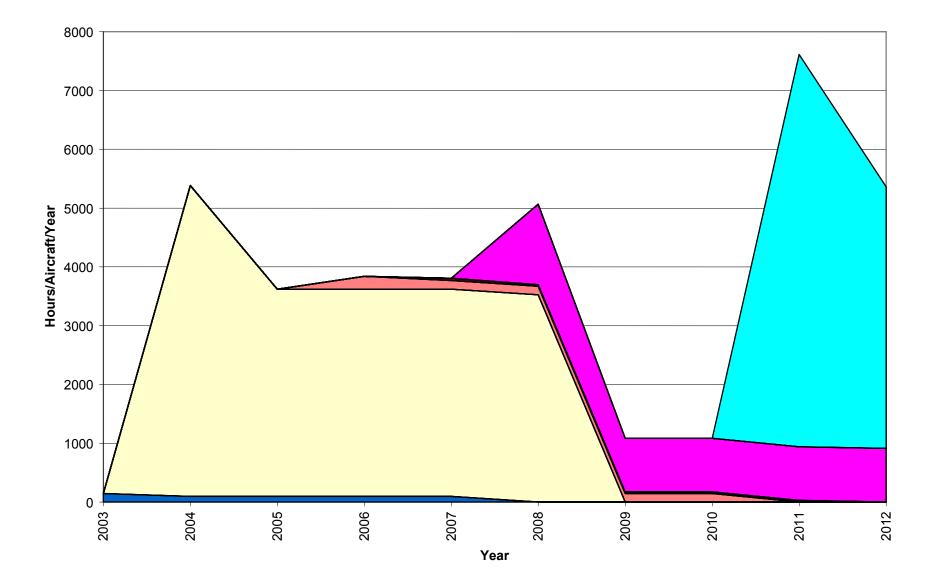


Typical Maintenance Issues



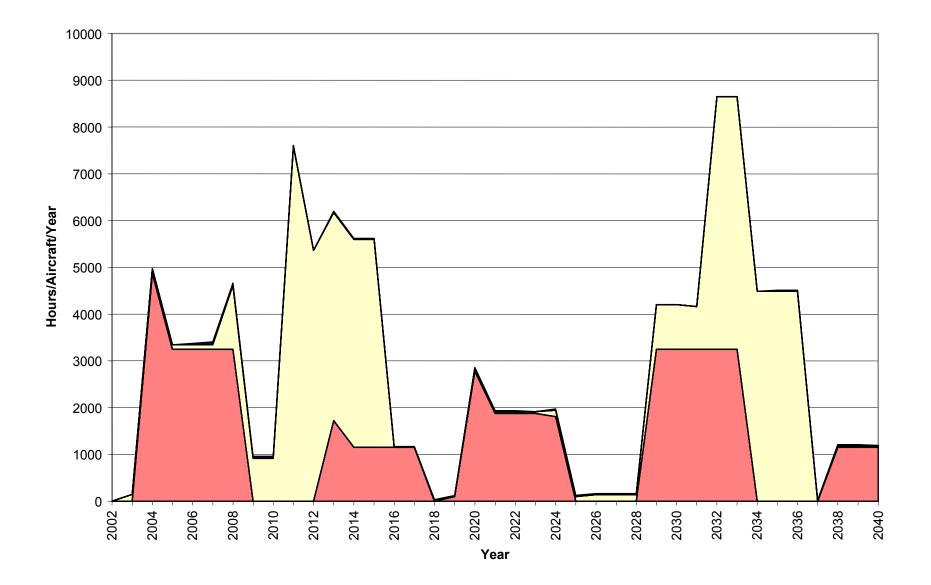


Near Term Inspections





Life Time Inspections

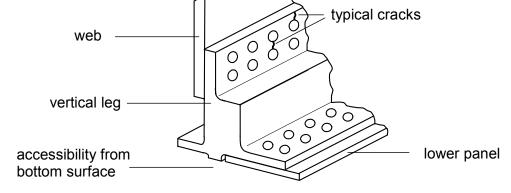




- Address High Costs for Performing Existing POD Evaluation
 - High cost of parts (material)
 - High cost of flaw creation

Ex: C-130 Beam Cap Holes

- Labor to perform POD study
- Additional Opportunities using Model-Assisted Approaches
 - Streamline validation of new technologies for in-field application
 - Improve confidence in NDE techniques for complex inspections
 - Address wide variations in flaw characteristics and location
 - Address variations in part geometry



(B1 wing carry through)

(corner cracks, alpha particles)



Prior POD Validation Studies

•	Have Cracks Will Travel (1979)	(crack detection)	
•	Retirement for Cause (RFC)	(crack detection)	
•	WRALC / SAIC Ultralmage Int. (Aging Aircraft Program Office) (1997 - 2004)		
	 C-141 Splice Joint 	(crack detection)	
	 C-141 Weep Hole 	(crack detection)	
	 C-130 Hat Section / C-130 Rainbow Fitting Holes 	(crack detection)	
	 C-130 Beam Cap Holes 	(crack detection)	
•	AFRL - Aging Aircraft Program Office / Sandia NL		
	 FastFocus system – RD Tech (2003) 	(crack detection)	
•	ACDP UDRI	(corrosion detection)	
•	Sandia NL Studies		
	 727 Fuselage Lap Joint Lower Skin 	(crack / corrosion detection)	

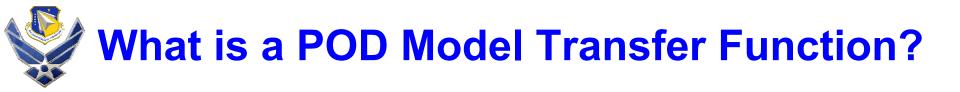


- Address transition of techniques to other aircraft (with varying part geometry and/or material properties)
 - from C-141, C-130, KC-135 etc.
 - to A-10, C-5, C-17 etc.
- Address costs for validation of new technologies
 - New sensors
 - EC: MWM, RFEC, GMR arrays
 - UT: Phased arrays (FastFocus, TESI program)
 - New techniques (Pulsed EC)

(Full POD validation exists for original part and technique)



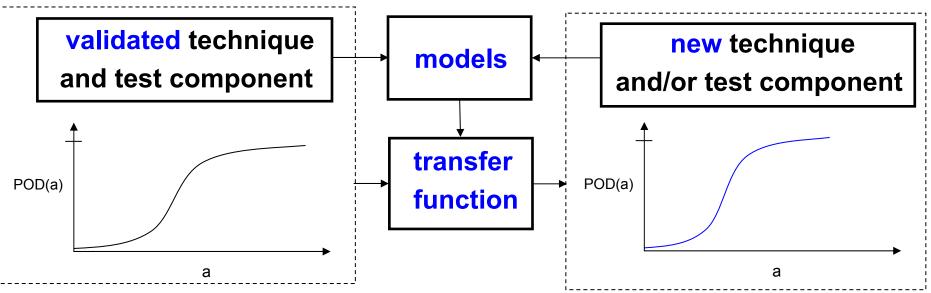
- Use lower cost manufactured flaws for full POD and extrapolate POD results for real flaws using accurate simulations and/or prior empirical data
 - EDM notches for real cracks
 - Simulated defects in engine components
- Reduce number of experimental samples required for a full POD and extrapolate POD results for real flaws using accurate simulations and/or prior empirical data



Approach: Extrapolated POD (M. Golis)

- Description:
 - POD results have been well established (RFC)
 - Minor changes in equipment (probes) or part geometry
 - Assess equivalent POD without need a full-scale evaluation

• Diagram:





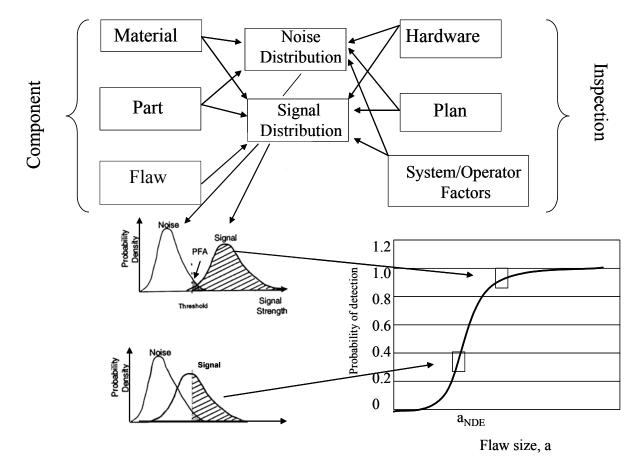
Approach: Extrapolated POD

- Potential methodology
 - Apply protocol to evaluate key parameters impacting NDE (NDE Insight, modify protocol for model-based evaluation)
 - Construct models for validated system and new system (system = technique and test component)
 - Evaluate model-based POD (for intrinsic capability with key application parameters) for both validated and new systems
 - Calculate transfer function between two model-based PODs
 - *linear transformation (?)*
 - nonlinear transformation (?)
 - Apply transfer function to original POD for validated system to estimate new system POD (incorporating human factors)



Approach: Modular POD (B. Thompson)

 Quantify signal and noise distributions using a modular assessment via simulated and experimental studies





Approach: Modular POD (B. Thompson)

- Methodology
 - Identify factors whose influence can be simulated using a physics based model
 - Develop appropriate model
 - Verify its accuracy in the laboratory through well controlled experiments
 - Use simulation tool to predict mean response and those components of variability controlled by well understood physical phenomena
 - Quantify additional sources of variability not controlled by well understood physical phenomena or associated with variations of input parameters that cannot be fully controlled in the production environment
 - Compute POD



Future Need for POD Determination (New POD Models)

- POD model relationships and validation studies for multiple quantitative measures to characterize a single flaw parameter
 - Operators use multiple features for making calls (C-scan, B-scan image data)
 - Automated Signal Classification also will take advantage of multiple features -> translate to final classification call
- POD model relationships and validation studies for multiple quantitative measures to characterize multiple flaw parameters
 - Corrosion (thickness loss, spatial extent, SCC, exfoliation)
 - Geometric flaws in engine components (3D POD)
- Validating NDE techniques with flaw classification procedures incorporating model-based inverse methods